

AESOP Field Prediction

by Brian B. Luu and Calvin D. Le

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1. Introduction

Under the sponsorship of the Defense Nuclear Agency (DNA), the Physics International Company developed a Transportable Electromagnetic Pulse Simulator (TEMPS) for Harry Diamond Laboratories (HDL)* in 1972. In 1974, HDL procured a copy of TEMPS, called AESOP (Army electromagnetic pulse (EMP) Simulator Operation), which was erected at the Woodbridge Research Facility (WRF) (see fig. 1). In 1994, AESOP was transferred to the Test and Evaluation Command (TECOM) at White Sands Missile Range. The main function of AESOP is to simulate the electromagnetic environment produced by an exoatmospheric nuclear burst. By simulating the threat-level EMP environment, it allows for testing and verifying the EMP susceptibility or hardness of a system. Since AESOP is a virtual copy of TEMPS (except for transportability and some minor modifications and enhancements), the specifications of AESOP are documented in the TEMPS's technical manuals.

Because of the complexity of EMP simulation and the limited computing capability in the 70s, no computing codes were developed to calculate or approximate the field generated by AESOP. A field mapping report¹ of TEMPS was developed to provide the EMP fields at key locations near the ground. The present report provides results from a code that calculates the electromagnetic (EM) field generated by AESOP (the code is given in app A).



Figure 1. AESOP at WRF.

^{*}Now part of the Army Research Laboratory.

¹Eugene L. Patrick and Spencer L. Soohoo, Transportable Electromagnetic Pulse Simulator (TEMPS) Field Mapping Report, Harry Diamond Laboratories, HDL-TR-1743 (Reprinted May 1984).

2. AESOP Simulator

Most of the information in this section is derived from the Physics International manual for TEMPS.²

Located at WRF in a 120,000-m² test area, AESOP is the largest EMP free-field simulator in the world. It consists of three basic subsystems: a support structure, pulse generator, and antenna.

- The support structure, made of fiberglass columns and laminated wood beams, is used to support and suspend the pulse generator and antenna at a height up to 20 m above the ground.
- The pulse generator mainly comprises the pulser assembly with two Marx generators, peaking capacitors, output switch, and control subsystem.
- The antenna is composed of a cylindrical and conical system of 36 wires extending from each end of the pulser assembly with associated forming hoops, end assemblies, tensioning elements, and end grounding terminations. The diameter of the cylindrical section is 9.144 m.

The general layout of the AESOP system is shown in figure 2.2 The overall length of AESOP is 300 m from end dome to end dome. The pulser and antenna are normally positioned at 20 m above the ground level but can also be at lower elevations.

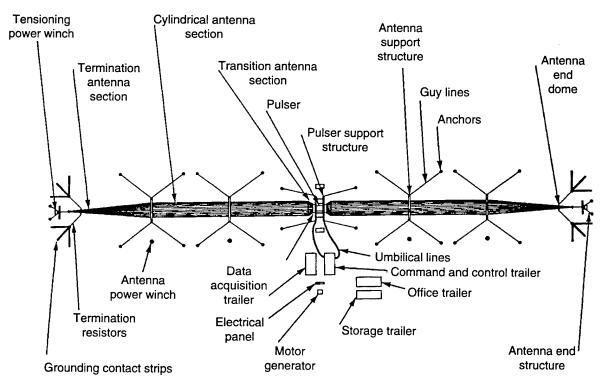
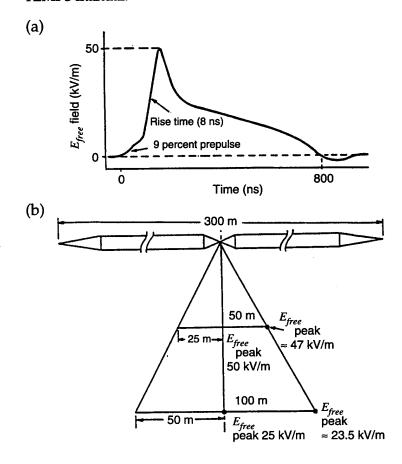


Figure 2. General layout of AESOP system.

²Physics International Company, Transportable Electromagnetic Pulse Simulator System (TEMPS) Operation & Maintenance Manual, PIMM-372 (April 1973).

AESOP produces a pulsed EM field by transmitting a high-energy electrical pulse generated from the pulse generator to the ground through the antenna. It can provide a radiated free-field range of 20 to 50 kV/m from the pulser on the center line. The pulse shape is a double exponential with less than 10-percent undershoot (fig. 3). For more information, see the TEMPS manual.²

Figure 3. AESOP characteristics: (a) pulse shape and (b) span on EMP coverage.



3. Theory

In the model presented here, the AESOP antenna is modeled as a series of dipole antenna elements above a perfectly conducting ground. There are two steps involved in the calculation of the radiated electromagnetic field from AESOP. A closed-form analytical time-domain current profile on each dipole element is derived by the transmission-line method. (In order to simplify the problem, we account only for reflection at the load end of AESOP.) The time-domain radiated electromagnetic field from each dipole is then computed by a convolution technique.

²Physics International Company, Transportable Electromagnetic Pulse Simulator System (TEMPS) Operation & Maintenance Manual, PIMM-372 (April 1973).

3.1 Current Distribution

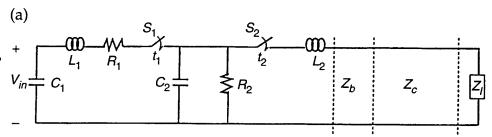
AESOP is represented by a transmission line over a perfectly conducting ground. There are three factors in this transmission-line model that need to be considered.²

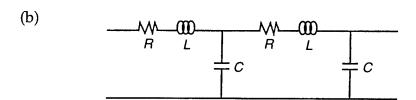
First, the 35-stage Marx generator can be modeled in a lumped equivalent circuit, as shown in figure 4(a). The switches S_1 and S_2 are the voltage charging and discharging switches, respectively. Z_a is the characteristic impedance of the antenna, which includes the biconical and conical sections. Z_1 is the termination load.

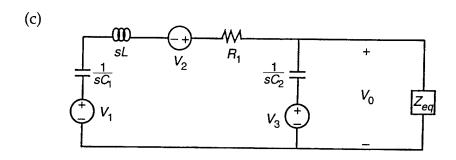
Second, the biconical section with cones at angle θ = 40.5° has a characteristic impedance, Z_b ,

$$Z_b = \frac{Z_0}{\pi} \ln \left[\cot \left(\frac{\theta}{2} \right) \right] \approx 120 \,\Omega$$
 (1)

Figure 4. Circuit models: (a) AESOP, (b) transmission-line model, and (c) AESOP model at $t = t_2$.







²Physics International Company, Transportable Electromagnetic Pulse Simulator System (TEMPS) Operation & Maintenance Manual, PIMM-372 (April 1973).

where

$$Z_0 = \sqrt{\frac{\mu_0}{\varepsilon_0}} \approx 377 \, \Omega \ .$$

Third, the cylindrical system of wires that makes up the AESOP antenna structure can be treated as low-loss transmission line of a wire over the ground plane. Its characteristic impedance, Z_c , is approximated by R, L, and C (R is series resistance per unit length, L is series inductance per unit length, and C is shunt capacitance per unit length):

$$Z_c = \left(1 + \frac{R}{2sL}\right)\sqrt{\frac{L}{C}} \approx 130\left(1 + \frac{R}{2sL}\right), \tag{2}$$

where

$$L = \frac{\mu_0}{2\pi} \ln\left(\frac{2h}{a}\right) ,$$

$$C = \frac{2\pi\varepsilon_0}{\ln\left(\frac{2h}{a}\right)} ,$$

 $s = j\omega$,

 ω = angular frequency,

h = height of a wire over the ground plane, and

a = radius of a wire.

The conical section is designed to maintain the same characteristic impedance as the cylindrical section. In other words, the ratio of 2h/a remains constant for both conical and cylindrical sections.

For the low-resistive-loss biconical section, the characteristic impedance, Z_b , is very compatible with the characteristic impedance, Z_c , of the cylindrical section. Since the cylindrical section is much longer than the biconical section, the assumption is that the characteristic impedance Z_b equals the characteristic impedance Z_c . The schematic representation of a 1-cm-unit-length section, δ_l , of Z_c is shown in figure 4(b).

At time $t = t_2$, we obtain the Laplace transform for the AESOP model (fig. 4(c)):

$$V_0 = V_3(t_2) \left(\frac{Z_l \sqrt{\frac{L}{C}}}{L_2 \left(Z_l + \sqrt{\frac{L}{C}} \right)} \right) \left(\frac{\prod_{i=1}^3 (s - y_i)}{\prod_{i=1}^5 (s - s_i)} \right) , \qquad (3)$$

where

 V_1 , V_2 , V_3 are initial voltage sources at $t = t_2^-$,

 Z_{eq} is the equivalent impedance of R_2 , L_2 , Z_b , Z_c , and Z_l ,

 y_1, y_2 are roots of

$$y^2 + \left[\frac{V_2(t_2)}{V_3(t_2) C_2} + \frac{R_1}{L_1} \right] y + \left[\frac{V_1(t_2)}{V_3(t_2) L_1 C_2} + \frac{1}{L_1 C_1} \right] = 0 \ ,$$

$$y_3 = \frac{R}{2L}$$
,

s₁, s₂, s₃, s₄, s₅ are roots of

$$s^5 + A_1 s^4 + A_2 s^3 + A_3 s^2 + A_4 s + A_5 = 0$$
, and

 A_1 , A_2 , A_3 , A_4 , A_5 are functions of R, R_1 , R_2 , L, L_1 , L_2 , C, C_1 , C_2 .

The reflecting voltage coefficient at the load end is

$$\Gamma = \frac{Z_l - Z_c}{Z_l + Z_c} = |\Gamma| e^{j\phi} . \tag{4}$$

Voltage across the capacitor, V_n , at any given n^{th} section of the antenna is calculated as

$$V_n = V_0 \left[e^{-n\psi} + \Gamma e^{-(2N-n)\psi} \right] , \qquad (5)$$

where

$$\psi = \alpha + j\beta ,$$

$$\alpha = \frac{R}{2\sqrt{\frac{L}{C}}} \delta l ,$$

$$\beta = \omega \sqrt{\frac{L}{C}} \delta l .$$

The corresponding current on the n^{th} section of the antenna is

$$I_n = (V_n - V_{n+1}) \frac{1}{L} \left(\frac{1}{s + \frac{R}{L}} \right) . \tag{6}$$

The time-domain current, $I_n(t)$, can be calculated directly from the above equation by the Laplace transform technique:

$$I_n(t) = \sum_{k=1}^{14} \left[B_{nk} \exp\left(D_{nk} t\right) \right] u(t - t_n) , \qquad (7)$$

where

 B_{nk} is the time constant coefficient number for the k^{th} variable and the n^{th} section,

 D_{nk} is the time constant exponential coefficient number for the k^{th} variable and the n^{th} section, and

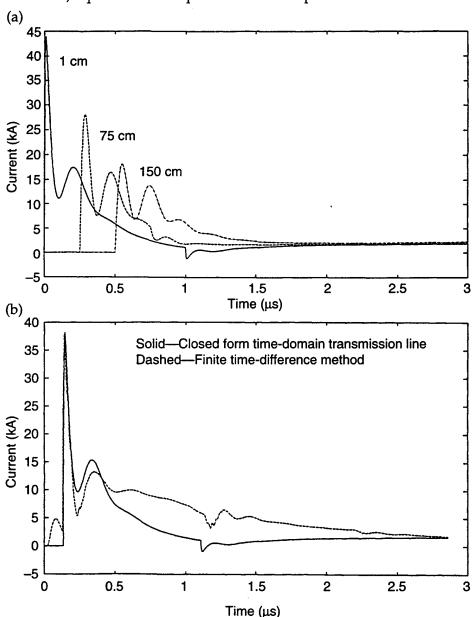
 t_n is the delay time constant for the n^{th} section.

The calculated current distributions at various locations on the AESOP antenna are shown in figure 5(a). Figure 5(b) compares the results of this method to those of the finite time-difference method. Good agreement was obtained at the early time, and reasonable agreement was obtained at the later time.

3.2 Electric Field and Magnetic Field of a Dipole

As described in section 2, an electromagnetic field is emitted from the AESOP antenna, which is a transmission-line antenna. It can be modeled as a series of linear dipoles. Each dipole itself is an elementary antenna that radiates fields. In this section, the electric field and magnetic field (radiated free field) equations of a dipole model are expanded from section 2.3 of

Figure 5. Current comparisons: (a) on different locations away from center of AESOP and (b) on bicone and at 1.3 ft away from center.



Collin³ and used as the basis for computing the fields of the AESOP simulator.

We begin with the hypothesis of a short current filament (dl), which can be represented as a dipole, located at the origin and oriented along the z-axis as shown in figure 6(a). The magnetic and electric fields generated by the dipole at an observation point are given by

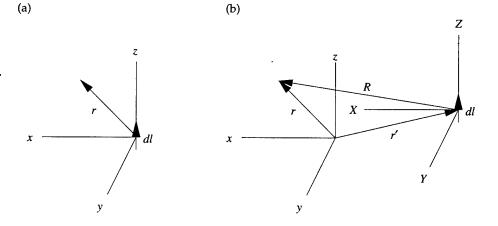
$$\overline{E} = \frac{Z_0 dl}{2\pi} \left(\frac{1}{r^2} + \frac{1}{jk_0 r^3} \right) I e^{-jk_0 r} \cos \theta \ \hat{r} + \frac{Z_0 dl}{4\pi} \left(\frac{jk_0}{r} + \frac{1}{r^2} + \frac{1}{jk_0 r^3} \right) I e^{-jk_0 r} \sin \theta \, \hat{\theta} \ , \ (8)$$

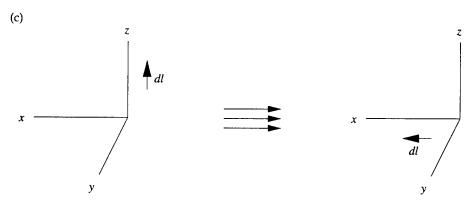
$$\overline{H} = \frac{dl}{4\pi} \left(\frac{jk_0}{r} + \frac{1}{r^2} \right) I e^{-jk_0 r} \sin \theta \ \hat{\phi} \quad , \tag{9}$$

where

 $Z_0 = (\mu_0/\varepsilon_0)^{1/2} \equiv$ intrinsic impedance of free space, $k_0 = \omega(\mu_0\varepsilon_0)^{1/2} \equiv$ free—space wave number, and $c = (\mu_0\varepsilon_0)^{-1/2} \equiv$ speed of light in free space.

Figure 6. Dipole model: (a) original, (b) general coordinates, and (c) rotate coordinates.





³Robert E. Collin, Antennas and Radiowave Propagation, McGraw-Hill Book Company, 1985.

In the more general case shown in figure 6(b), r(x,y,z) and r'(x',y',z') are the coordinates of the observation point and of the dipole, respectively. Hence, R(X,Y,Z) is the relative coordinate of the observation point in the Cartesian coordinate with the origin at the dipole, where $\vec{R} = \vec{r} - \vec{r}'$. The magnitude of R can be defined as

$$R = \sqrt{(x-x')^2 + (y-y')^2 + (z-z')^2}.$$

In the general case, then, equations (8) and (9) become

$$\overline{E}(\omega) = \frac{Z_0 dl}{2\pi} \left(\frac{1}{R^2} + \frac{c}{j\omega R^3} \right) I(\omega) e^{-j\omega R/c} \cos \theta \, \hat{R}
+ \frac{Z_0 dl}{4\pi} \left(\frac{j\omega}{cR} + \frac{1}{R^2} + \frac{c}{j\omega R^3} \right) I(\omega) e^{-j\omega R/c} \sin \theta \, \hat{\Theta} ,$$
(10)

$$\vec{H}(\omega) = \frac{dl}{4\pi} \left(\frac{j\omega}{cR} + \frac{1}{R^2} \right) I(\omega) e^{-j\omega R/c} \sin \theta \,\hat{\Phi} . \tag{11}$$

By transferring equations (10) and (11) into the time domain and changing to Cartesian coordinates, we transform equations (10) and (11) to

$$\vec{E}(t) = \frac{Z_0 dI}{4\pi} \left[\frac{1}{cR} \frac{\partial I}{\partial t} \left(t - \frac{R}{c} \right) + \frac{3}{R^2} I \left(t - \frac{R}{c} \right) + \frac{3c}{2R^3} \operatorname{sgn}(t) \otimes I \left(t - \frac{R}{c} \right) \right] \frac{XZ}{R^2} \hat{x}
+ \frac{Z_0 dI}{4\pi} \left[\frac{1}{cR} \frac{\partial I}{\partial t} \left(t - \frac{R}{c} \right) + \frac{3}{R^2} I \left(t - \frac{R}{c} \right) + \frac{3c}{2R^3} \operatorname{sgn}(t) \otimes I \left(t - \frac{R}{c} \right) \right] \frac{YZ}{R^2} \hat{y}
+ \frac{Z_0 dI}{4\pi} \left(\frac{1}{cR} \frac{\partial I}{\partial t} \left(t - \frac{R}{c} \right) \frac{Z^2 - R^2}{R^2} + \left[\frac{1}{R^2} I \left(t - \frac{R}{c} \right) + \frac{c}{2R^3} \operatorname{sgn}(t) \otimes I \left(t - \frac{R}{c} \right) \right] \frac{3Z^2 - R^2}{R^2} \hat{z} ,$$
(12)

$$\overline{H}(t) = \frac{dl}{4\pi} \left[\frac{1}{cR} \frac{\partial I}{\partial t} \left(t - \frac{R}{c} \right) + \frac{1}{R^2} I \left(t - \frac{R}{c} \right) \right] \left(-\frac{Y}{R} \hat{x} + \frac{X}{R} \hat{y} \right) , \tag{13}$$

where $\operatorname{sgn}(t) = \frac{t}{|t|}$, $t \neq 0$, and \otimes is a convolution operator.

We can change the orientation of the dipole so that it matches those of AESOP by rotating the coordinates as indicated in figure 6(c):

from
$$x$$
 to y
 y to z
 z to x

Finally, the electric and magnetic fields of a dipole are calculated by

$$\vec{E}(t) = \frac{Z_0 dl}{4\pi} \left(\frac{1}{cR} \frac{\partial I}{\partial t} \left(t - \frac{R}{c} \right) \frac{X^2 - R^2}{R^2} + \left[\frac{1}{R^2} I \left(t - \frac{R}{c} \right) + \frac{c}{2R^3} \operatorname{sgn}(t) \otimes I \left(t - \frac{R}{c} \right) \right] \frac{3X^2 - R^2}{R^2} \right) \hat{x}$$

$$+ \frac{Z_0 dl}{4\pi} \left[\frac{1}{cR} \frac{\partial I}{\partial t} \left(t - \frac{R}{c} \right) + \frac{3}{R^2} I \left(t - \frac{R}{c} \right) + \frac{3c}{2R^3} \operatorname{sgn}(t) \otimes I \left(t - \frac{R}{c} \right) \right] \frac{XY}{R^2} \hat{y}$$

$$+ \frac{Z_0 dl}{4\pi} \left[\frac{1}{cR} \frac{\partial I}{\partial t} \left(t - \frac{R}{c} \right) + \frac{3}{R^2} I \left(t - \frac{R}{c} \right) + \frac{3c}{2R^3} \operatorname{sgn}(t) \otimes I \left(t - \frac{R}{c} \right) \right] \frac{XZ}{R^2} \hat{z} , \tag{14}$$

$$\overline{H}(t) = \frac{dI}{4\pi} \left[\frac{1}{cR} \frac{\partial I}{\partial t} \left(t - \frac{R}{c} \right) + \frac{1}{R^2} I \left(t - \frac{R}{c} \right) \right] \left(-\frac{Z}{R} \hat{y} + \frac{Y}{R} \hat{z} \right) . \tag{15}$$

4. Implementation

The dipole current distribution and field equations developed in section 3 provide us the means to compute the free field emitted by AESOP. But the total field in the half space above the ground is a result of the superposition of the free field and the reflected field from the ground.

To simplify the problem, we assume that the earth ground is a perfectly electrically conducting (PEC) half space. With this assumption, we can apply the method of images to calculate the reflected field. The calculation can be further simplified if we use the symmetry of dipole currents through the plane that is perpendicular to the axis of AESOP at the center. The biconic and cylindrical parts of the antenna are assumed to be one single transmission line, having both ends slanted toward the ground.

We model the simulator as a series of dipole elements representing the physical structure of the AESOP simulator. The number of dipole elements has been selected to resolve the early-time structure of the field and to produce a relatively smooth waveform. For every 0.01-m section of the antenna, we evaluate the current distribution developed in section 3.1 at the time and spatial location that is the driving source for the corresponding dipole element. We employ 14,978 dipole elements with a dipole separation of 0.01 m to represent half the antenna of AESOP.

We let the coordinates be as depicted in figure 7. Thus, the z=0 plane is the ground plane and parallel to AESOP. The method of images is applied through the ground plane to compute the reflected fields. The y=0 plane contains the antenna. The symmetry of current distribution is through the x=0 plane, called the center plane. The height of the antenna is 20 m above the ground. Each end of AESOP is terminated at 6.5 m above the ground. Each slanted section begins and ends at 100.4 and 147.9 m from the center plane, respectively. To save computational time, we translate and rotate the coordinate system to determine the fields of the simulator slanted sec-

tions, as depicted in figure 8. Finally, we use superposition of all fields to obtain the total field at an observation point from the AESOP simulator.

We have used a FORTRAN program to compute all fields at an observation point. The program is listed in appendix A.

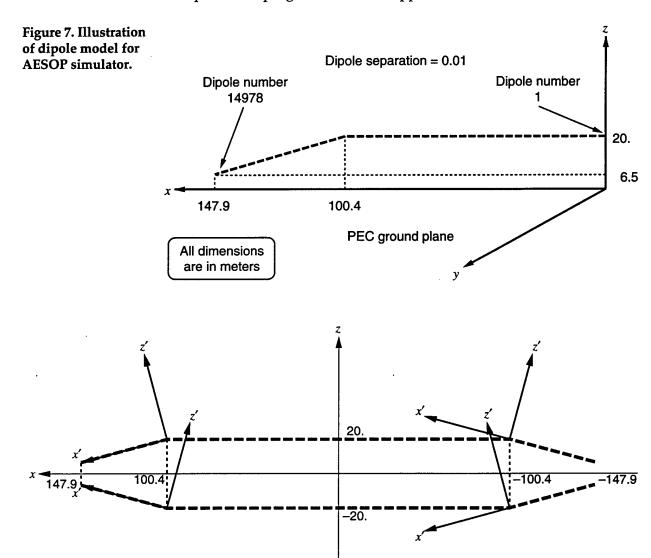


Figure 8. Coordinates for slanted dipoles.

5. Results and Comparison with Empirical Data

We implemented a FORTRAN program to compute the fields at an observation point out to $2 \mu s$ in time with a time step of 1 ns. The following are the execution times of the program on some systems:

System	Execution time (hour:minute)
IBM 3090 with MVS/XA 2.3	4:00
IBM RISC/6000-530 with AIX 3.2.4	3:00
IBM RISC/6000-550 with AIX 3.2.4	2:18
IBM RISC/6000-560 with AIX 3.2.4	1:50
SGI 4D/440 with IRIX 4.0.5	2:32
SGI 4D/35 with IRIX 4.0.5	11:11
DECstation 5000 with ULTRIX 4.3	14:49

Because of the symmetry, the computed field components E_x , H_y , and H_z are the only nonzero components of the field on the center plane (x=0 plane). Off the center plane, the computed model indicates that the H_x component is zero (in retarded time) until the current arrives to the slanted section of the simulator. In the plane containing the antenna (y=0 plane), the symmetry requires that E_x , E_z , and H_y be the only nonzero fields. The calculated fields show the effects of the change in direction of the current on the antenna structure, as slope discontinuities in the fields, at the expected time.

For comparison to the empirical data, we scale all the computed fields by a factor that scales the computed total electric field of the point (0,50,20) to 50 kV/m. In comparison, we notice that the idealized lossless ground plane assumption has caused the peak fields of the model to be generally overestimated (compared to the empirical data) by roughly a factor of two; this assumption also led to the inaccuracy of the vertical field prediction from the slanted end sections of the simulator. Assuming a perfectly conducting ground plane means that the model does not reflect the effects of the pseudo-Brewster angle on the vertical fields. However, the energy in the predicted field waveforms still provides an upper bound for the actual fields.

Appendix B contains field plots compared to empirical data for the following sample points in the space:

```
( 0, -176, 1)
( 0, -176, 4.6)
(-332, -176, 1)
(-332, -176, 4.6)
( 0, 1546, 4.6)
( 410, 502, 4.6)
(-444, 1341, 4.6)
( 473, 0, 1)
( 473, 0, 4.6)
```

6. Conclusion

Although the perfectly electrically conducting ground assumption has generally overestimated and distorted the calculated field, the predicted fields are still good for bounding problems, especially for points high above the ground. The calculated information can provide upper bound contours or isosurfaces at certain field magnitudes. The number of dipoles used in the model can be decreased, with a corresponding decrease in the model run time, but with an increase in "spiky" noise in the predicted waveforms. This model has been helpful in understanding the relationship between the characteristics of the field waveforms and the structure of the AESOP simulator. Because it is simple and requires only 2.5 MB of memory to execute, this model can be used on most computer platforms. This code is the first building block to develop more accurate field predictions for the AESOP simulator.

Appendix A.—Program Listing

С

С

```
IMPLICIT NONE
   INTEGER*4 MAXF, MAXS, MAXT, MAXSH
   PARAMETER (MAXF=8, MAXS=14978, MAXSH=10040, MAXT=2000)
   CHARACTER DSCUCO*80, DSEHD*80, DSEHT*80, SPC*1
   REAL*8 COEF(20, MAXS), CUC
   REAL*8 CUCC, CUL, CUR
   REAL*8 EH(0:MAXT,MAXF), RL(4), RO(12), RS(4)
   REAL*8 EHM(8), ROS(4), ROT(4)
   REAL*8 OXT, OZT, SA, SAX, SAZ, SINA, COSA
   REAL*8 AL, C, DL, DLMIN, DT, PI, ZO, ZOO4PI
   REAL*8 CU, CUD, CVO, ETC, HTC, TCS, TEHS, TR, RETARD
   REAL*8 CUP, TCRS, SN, SN1, SNR, SNR1
   INTEGER*4 IB, INF, IT, ITS, IS, ISB, ISE, ISL, ISGX, ISGZ
   INTEGER*4 I, J, K, ITM(8), ISGXB, ISGXE, ISGZB, ISGZE
   COMMON /COEFBL/COEF, CUC
   COMMON /CUPARM/CUP, TCRS, SN, SN1, SNR, SNR1
  DATA CUCC, CUL, CUR/0.273448D-12,4.07D-9,5.364D-3/
   CUP = DSQRT(CUL*CUCC)
   CUC = CUR/(2.*DSQRT(CUL/CUCC))
INITIALIZE PARAMETERS
   ISGXB = 1
   ISGXE = -1
  ISGZB = 1
  ISGZE = -1
  SPC = '^'
  C = 2.997925D8
  C = 3.D8
  PI = 4.*DATAN(1.D0)
  ZO = 4.*PI*C*1.D-7
  ZO = 376.991184D0
  ZOO4PI=C*1.D-7
  DLMIN = 1.D-2
  DT = 1.D-9
  AL = 20.
  RS(2) = 0.
  OXT = 100.40D0
  OZT = AL
  SAX = 47.5D0
  SAZ = 13.5D0
  SA = DSQRT(SAX**2 + SAZ**2)
```

```
COSA = SAX/SA
      SINA = SAZ/SA
      ISE = 0
  READ IN CURRENT COEFFICIENTS OF ALL SEGMENTS
      DSCUCO = '/home/emsc3/bluu/aesop/CUCOEF.DATA'
      OPEN(28, FILE=DSCUCO, STATUS='OLD', FORM='FORMATTED')
      DO 170 J=1, MAXS
         READ(28,*) IS, (COEF(I,J), I=1,20)
  170 CONTINUE
      CLOSE(28)
* READ PARTITION DATA SET NAME OF THE E AND H FIELD
      READ(5, '(A)') DSEHD
      IB = INDEX(DSEHD, ' ')
      IB = IB - 1
 READ IN SEGMENT LENGTH AND COORDINATE OF OBSERVATION POINT
  115 READ(5, *, END=999) DL, (RO(I), I=1,12)
      ISL = (DL + DLMIN/2)/DLMIN
 DETERMINE THE STARTING TIME
      ISB = (ISL+1)/2
      ISGX = 1
      IF (RO(1).LT.0.) ISGX = -1
     RS(1) = ISGX*(2*ISB-1)*(DLMIN/2.)
     TEHS = DSQRT((RO(1)-RS(1))**2 + RO(2)**2 + (RO(3)-AL)**2)/C +
     ISB*CUP
 INITIALIZE EH ARRAY
     DO 150 J=1, MAXF
        DO 150 I=0, MAXT
           EH(I,J) = 0.
 150 CONTINUE
* CALCULATE E AND H FIELD
     WRITE(6,*) 'P', RO(1), RO(2), RO(3)
     WRITE(6,*) ' DIPOLE LENGTH IS', DL
```

```
DO 200 IS=ISB, MAXSH, ISL
         ISE = IS
         TCS = IS*CUP
         TCRS = (2*(MAXS-IS)+1)*CUP
         SN = IS-1
         SN1 = IS
         SNR = 2*MAXS - IS
         SNR1 = 2*MAXS - IS +1
         DO 220 ISGZ= ISGZB, ISGZE, -2
         DO 220 ISGX= ISGXB, ISGXE, -2
            RS(1) = ISGX*(2*IS-1)*(DLMIN/2.)
            RS(3) = ISGZ*AL
            RS(4) = DSQRT(RS(1)**2 + RS(2)**2 + RS(3)**2)
С
            RL(1) = RO(1) - RS(1)
            RL(2) = RO(2) - RS(2)
            RL(3) = RO(3) - RS(3)
            RL(4) = DSORT(RL(1)**2 + RL(2)**2 + RL(3)**2)
            RETARD = RL(4)/C + TCS - TEHS
            ITS = RETARD/DT + 1.
            DO 250 IT=ITS, MAXT
               TR = IT*DT - RETARD
               CALL CURRENT(IS, TR, CU, CUD, CVO)
               HTC = ISGZ*DL*(CUD/RL(4)/C +
                CU/RL(4)**2)/(4.*PI)
               EH(IT,4) = EH(IT,4) + 0.
               EH(IT,5) = EH(IT,5) - RL(3)/RL(4)*HTC
               EH(IT,6) = EH(IT,6) + RL(2)/RL(4)*HTC
               ETC = ISGZ*ZOO4PI*DL*(CUD/RL(4)/C +
                3.*(CU/RL(4)**2) + (1.5*C)*(CVO/RL(4)**3))
               EH(IT,1) = EH(IT,1) + ISGZ*ZOO4PI*DL*(
                CUD/RL(4)/C*((RL(1)/RL(4))**2-1.) +
                (CU/RL(4)**2+(C/2.)*(CVO/RL(4)**3))*
                (3.*(RL(1)/RL(4))**2-1.)
               EH(IT,2) = EH(IT,2) + RL(1)/RL(4)*RL(2)/RL(4)*ETC
               EH(IT,3) = EH(IT,3) + RL(1)/RL(4)*RL(3)/RL(4)*ETC
  250
            CONTINUE
  220
         CONTINUE
  200 CONTINUE
  CALCULATE THE FIELD CONTRIBUTE BY THE SLANT PARTS
      RS(3) = 0.
      ROT(2) = RO(2)
      ISB = ISE + ISL
```

```
DO 260 IS=ISB, MAXS, ISL
         TCS = IS*CUP
         TCRS = (2*(MAXS-IS)+1)*CUP
         SN = IS-1
         SN1 = IS
         SNR = 2*MAXS - IS
         SNR1= 2*MAXS - IS +1
         DO 280 ISGZ= ISGZB, ISGZE, -2
         DO 280 ISGX= ISGXB, ISGXE, -2
            ROS(1) = RO(1) - ISGX*OXT
            ROS(3) = RO(3) - ISGZ*OZT
            ROT(1) = COSA*ROS(1) - ISGX*ISGZ*SINA*ROS(3)
            ROT(3) = ISGX*ISGZ*SINA*ROS(1) + COSA*ROS(3)
            RS(1) = ISGX*(2*(IS-MAXSH)-1)*(DLMIN/2.)
C
            RS(4) = DSQRT(RS(1)**2 + RS(2)**2 + RS(3)**2)
            RL(1) = ROT(1) - RS(1)
            RL(2) = ROT(2) - RS(2)
            RL(3) = ROT(3) - RS(3)
            RL(4) = DSQRT(RL(1)**2 + RL(2)**2 + RL(3)**2)
            RETARD = RL(4)/C + TCS - TEHS
            ITS = RETARD/DT + 1.
            DO 290 IT=ITS, MAXT
               TR = IT*DT - RETARD
               CALL CURRENT(IS, TR, CU, CUD, CVO)
               HTC = ISGZ*DL*(CUD/RL(4)/C +
               CU/RL(4)**2)/(4.*PI)
               EHM(6) = RL(2)/RL(4)*HTC
               EH(IT,4) = EH(IT,4) + ISGX*ISGZ*SINA*EHM(6)
               EH(IT,5) = EH(IT,5) - RL(3)/RL(4)*HTC
               EH(IT,6) = EH(IT,6) + COSA*EHM(6)
               ETC = ISGZ*ZOO4PI*DL*(CUD/RL(4)/C +
                3.*(CU/RL(4)**2) + (1.5*C)*(CVO/RL(4)**3))
               EHM(1) = ISGZ*ZOO4PI*DL*(
               CUD/RL(4)/C*((RL(1)/RL(4))**2-1.) +
                (CU/RL(4)**2+(C/2.)*(CVO/RL(4)**3))*
                (3.*(RL(1)/RL(4))**2-1.)
                                                           )
              EHM(3) = RL(1)/RL(4)*RL(3)/RL(4)*ETC
              EH(IT,1) = EH(IT,1) + COSA*EHM(1) + ISGX*ISGZ*SINA*EHM(3)
              EH(IT,2) = EH(IT,2) + RL(1)/RL(4)*RL(2)/RL(4)*ETC
              EH(IT,3) = EH(IT,3) - ISGX*ISGZ*SINA*EHM(1)+COSA*EHM(3)
```

```
290
          CONTINUE
280
      CONTINUE
260 CONTINUE
   DO 270 I=1,8
       EHM(I) = 0.
       ITM(I) = 0.
270 CONTINUE
DETERMINE THE MAX VALUES OF E & H FIELDS AND THEIR TIMES
   DO 295 IT=0, MAXT
      ETC = DSQRT(EH(IT,1)**2 + EH(IT,2)**2 + EH(IT,3)**2)
      HTC = DSQRT(EH(IT, 4)**2 + EH(IT, 5)**2 + EH(IT, 6)**2)
      EH(IT,7)=ETC
      EH(IT,8)=HTC
      IF (ETC.GT.EHM(7)) THEN
          EHM(7) = ETC
          ITM(7) = IT
      ENDIF
      IF (HTC.GT.EHM(8)) THEN
          EHM(8) = HTC
        II = IT
      ENDIF
       DO 297 I=1,6
          IF (ABS(EH(IT,I)).GT.ABS(EHM(I))) THEN
             EHM(I) = EH(IT,I)
             ITM(I) = IT
          ENDIF
297
     CONTINUE
295 CONTINUE
SET A DATA SET NAME FOR E AND H FIELD
   WRITE (DSEHT, '(SP, A, E14.7, E14.7, E14.7)') DSEHD(1:IB),
   RO(1), RO(2), RO(3)
   IF(DSEHT(IB+1:IB+1).NE.'') THEN
      DSEHT(IB+2:IB+2) = DSEHT(IB+1:IB+1)
      DSEHT(IB+16:IB+16) = DSEHT(IB+15:IB+15)
      DSEHT(IB+30:IB+30) = DSEHT(IB+29:IB+29)
   ENDIF
   DSEHT(IB+1:IB+1) = SPC
   DSEHT(IB+15:IB+15) = SPC
   DSEHT(IB+29:IB+29) = SPC
```

```
RETARD = TEHS/DT
   WRITE E AND H FIELD TO FILE
      OPEN(32, FILE=DSEHT, STATUS='UNKNOWN', FORM='UNFORMATTED')
      WRITE(32) RO(1),RO(2),RO(3),RO(4),RO(5),RO(6),DL,DT
      WRITE(32) RO(7), RO(8), RO(9), RO(10), RO(11), RO(12), TEHS, RETARD
      WRITE(32) EHM(1), EHM(2), EHM(3), EHM(4), EHM(5), EHM(6), EHM(7),
      EHM(8)
      WRITE(32) ITM(1), ITM(2), ITM(3), ITM(4), ITM(5), ITM(6),
      | ITM(7), ITM(8)
      DO 400 I=0, MAXT
         WRITE(32) (EH(I,J), J=1, 8)
  400 CONTINUE
      CLOSE (32)
      WRITE(6,*) 'END OF THIS RUN; DATA IN FILE: ', DSEHT
      GOTO 115
  999 STOP
   79 FORMAT(8E15.7)
      END
    SUBROUTINE CALCULATE THE DIPOLE'S
                          CURRENT,
                          CURRENT DERIVATIVE,
                          CONVOLUTION OF SIGN FUNCTION AND CURRENT
      SUBROUTINE CURRENT(NS, TS, Y, YD, YCO)
      IMPLICIT NONE
      INTEGER*4 MAXS, NS, LR
      REAL*8 P, TC2, SN, SN1, SNR, SNR1
      REAL*8 T, TS, Y, YS, YD, YSD, YCO, YSCO
      COMMON /CUPARM/P, TC2, SN, SN1, SNR, SNR1
С
      DATA P/0.333606558688524581D-10/
С
      DATA C/0.219836066437990879D-04/
      LR = 0
      T = TS
      CALL CURDIS (NS, LR, SN1, T, YS, YSD, YSCO)
      T = TS + P
     CALL CURDIS (NS, LR, SN, T, Y, YD, YCO)
     Y = Y - YS
```

```
YD = YD - YSD
   YCO = YCO - YSCO
REFLECTION CURRENT IS NOT EMERGING
   IF (T.LE.TC2) RETURN
   T = TS - TC2
REFLECTION CURRENT
   LR = 12
   CALL CURDIS (NS, LR, SNR1, T, YS, YSD, YSCO)
   Y = Y + YS
   YD = YD + YSD
   YCO = YCO + YSCO
   T = T + P
   CALL CURDIS (NS, LR, SNR, T, YS, YSD, YSCO)
   Y = Y - YS
   YD = YD - YSD
   YCO = YCO - YSCO
   RETURN
   END
 SUBROUTINE CALCULATES CURRENT PARAMETERS
   SUBROUTINE CURDIS (NS, IR, SN, T, YS, YSD, YSCO)
   IMPLICIT NONE
   INTEGER*4 MAXS, NS, IR
   PARAMETER (MAXS=14978)
   REAL*8 CO(20, MAXS), C
   REAL*8 T, SN, YS, YSD, YSCO
   COMMON /COEFBL/CO,C
   AA = CO(1+IR,NS)
   AB = CO(2+IR,NS)
   AC = CO(3+IR,NS)
   AD = CO(4+IR,NS)
   PD = CO(5+IR,NS)
   AF = CO(6+IR,NS)
   EA = CO(7,NS)
   EB = CO(8,NS)
   EC = CO(9,NS)
```

ED = CO(10,NS)

```
EDP = CO(11,NS)
   EF = CO(12,NS)
   RA = CO(13, NS)
   RB = CO(14,NS)
   RC = CO(15, NS)
   RD = CO(16,NS)
   PDR = CO(17,NS)
   RF = CO(18, NS)
   AR = CO(19, NS)
   ER = CO(20,NS)
CURRENT DISTRIBUTION
   YS = DEXP(-SN*C)*(CO(1+IR,NS)*DEXP(CO(7,NS)*T) +
                      CO(2+IR,NS)*DEXP(CO(8,NS)*T) +
                      CO(3+IR,NS)*DEXP(CO(9,NS)*T) +
                      CO(6+IR,NS)*DEXP(CO(12,NS)*T) +
  2.*CO(4+IR,NS)*DEXP(CO(10,NS)*T)*DCOS(CO(11,NS)*T+CO(5+IR,NS)))
DERIVATIVE OF CURRENT DISTRIBUTION
   YSD = DEXP(-SN*C)*(CO(1+IR,NS)*CO(7,NS)*DEXP(CO(7,NS)*T) +
                       CO(2+IR,NS)*CO(8,NS)*DEXP(CO(8,NS)*T) +
                       CO(3+IR,NS)*CO(9,NS)*DEXP(CO(9,NS)*T) +
                       CO(6+IR,NS)*CO(12,NS)*DEXP(CO(12,NS)*T) +
  2.*CO(4+IR,NS)*DEXP(CO(10,NS)*T)*(
                   CO(10, NS) *DCOS(CO(11, NS) *T+CO(5+IR, NS))-
                   CO(11,NS)*DSIN(CO(11,NS)*T+CO(5+IR,NS))))
CONVOLUTION OF SGN(T) AND I(T)
   YSCO = DEXP(-SN*C)*(
       CO(1+IR,NS)*(2.*DEXP(CO(7,NS)*T)-1.)/CO(7,NS) +
       CO(2+IR,NS)*(2.*DEXP(CO(8,NS)*T)-1.)/CO(8,NS) +
       CO(3+IR,NS)*(2.*DEXP(CO(9,NS)*T)-1.)/CO(9,NS) +
       CO(6+IR,NS)*(2.*DEXP(CO(12,NS)*T)-1.)/CO(12,NS) +
     2.*CO(4+IR,NS)*( 2.*DEXP(CO(10,NS)*T)*
                      (CO(10,NS)*DCOS(CO(11,NS)*T+CO(5+IR,NS))+
                       CO(11,NS)*DSIN(CO(11,NS)*T+CO(5+IR,NS))) -
                      (CO(10,NS)*DCOS(CO(5+IR,NS))+
                       CO(11,NS)*DSIN(CO(5+IR,NS))) )/
                    (CO(10,NS)**2+CO(11,NS)**2) )
  IF(IR.EQ.0) RETURN
  YS = YS + DEXP(-SN*C)*CO(19,NS)*DEXP(CO(20,NS)*T)
```

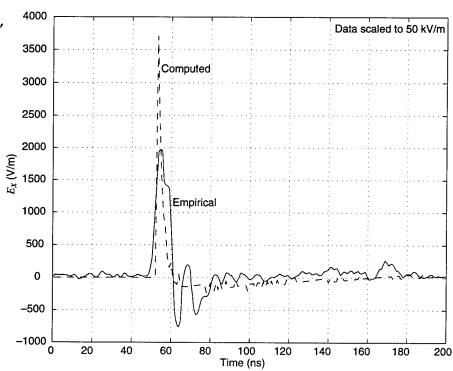
```
YSD= YSD + DEXP(-SN*C)*CO(19,NS)*CO(20,NS)*DEXP(CO(20,NS)*T)
YSCO= YSCO + DEXP(-SN*C)*CO(19,NS)*(2.*DEXP(CO(20,NS)*T)-1.)/
CO(20,NS)
RETURN
END
```

Appendix B.—Field Plot Comparison

The field plots presented display electric or magnetic field component waveforms at selected test points, showing both empirical measurements and computed results in the same graph. No comparisons are done for field components that are zero in theory. Both of the fields, empirical and computed, are scaled to the same reference field level, 50 kV/m at 50 m away from the center of AESOP on the center plane. We selected the field points based on the availability of empirical data for test points that were measured for the Perimeter Field Map at the Woodbridge Research Facility. For those test points that have more than one measurement, we have chosen the best data set to compare with computed data. The selected field comparison points are as follows:

```
0, -176,
                        (fig. B-1)
              1)
   0, -176, 4.6
                        (fig. B-2)
(-332, -176,
                       (fig. B-3)
               1)
(-322, -176, 4.6)
                       (fig. B-4)
   0, 1546, 4.6)
                       (fig. B-5)
                        (fig. B-6)
(410, 502, 4.6)
(-444, 1341, 4.6)
                       (fig. B-7)
(473,
                       (fig. B-8)
          0, 1)
(473,
          0, 4.6)
                       (fig. B-9)
```

Figure B-1. Field (a) comparison at (0,-176,1) meters: (a) E_x , (b) H_y .



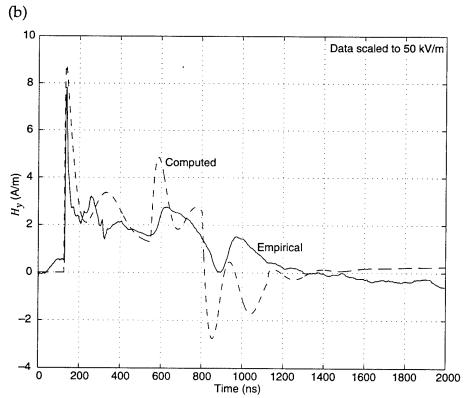


Figure B-1 (cont'd). Field comparison at (0,-176,1) meters: $(c) H_z$.

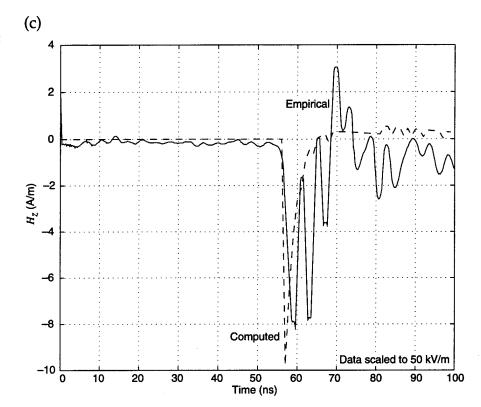
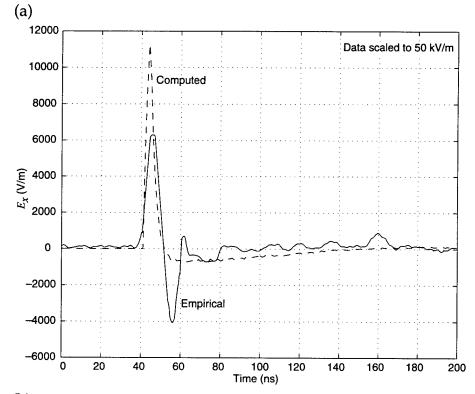


Figure B-2. Field comparison at (0,-176,4.6) meters: (a) E_x , (b) H_y .



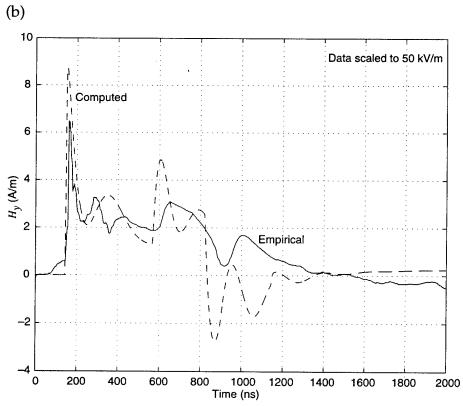


Figure B-2 (cont'd). Field comparison at (0,-176,4.6) meters: $(c) H_z$.

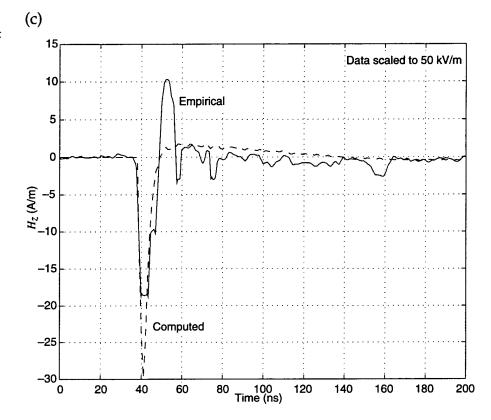
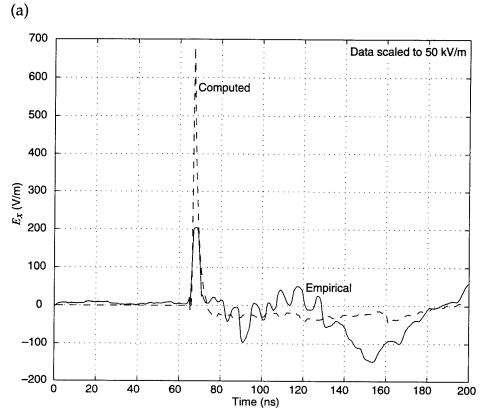


Figure B-3. Field comparison at (-332,-176,1) meters: (a) E_x , (b) E_y .



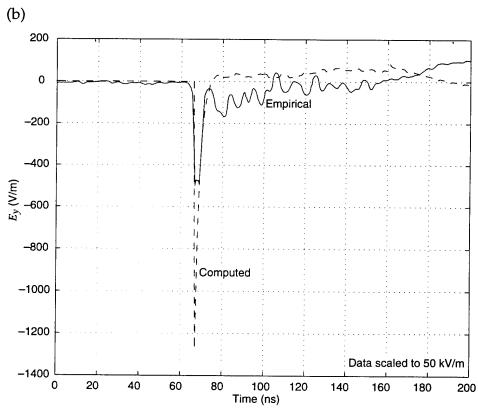
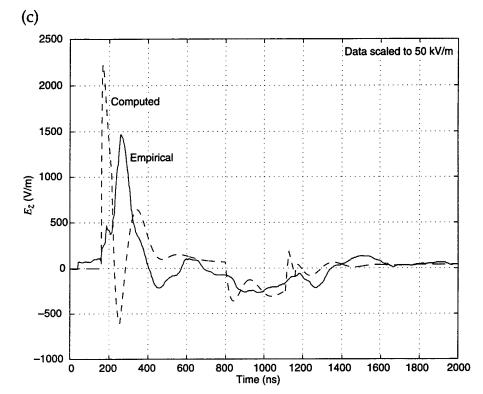


Figure B-3 (cont'd). Field comparison at (-332,-176,1) meters: (c) E_z , (d) H_x .



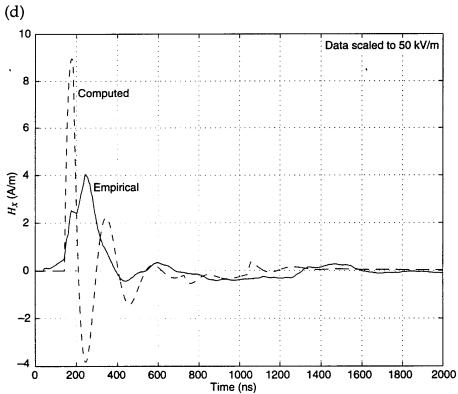
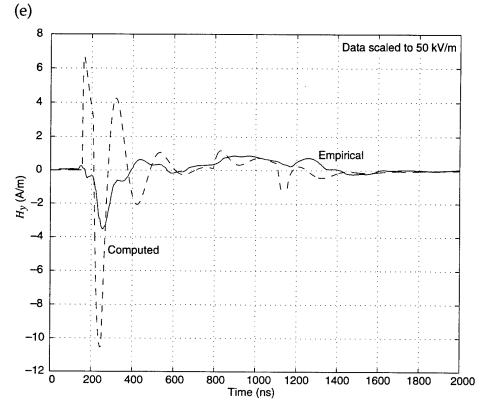


Figure B-3 (cont'd). Field comparison at (-332,-176,1) meters: (e) H_y , (f) H_z .



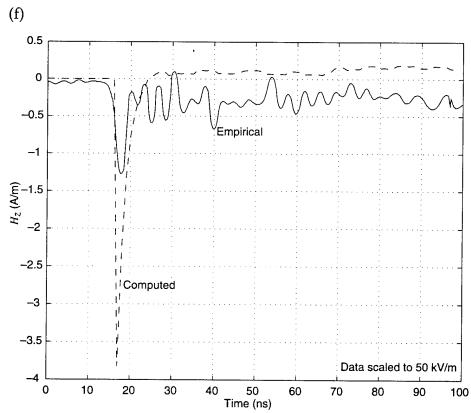
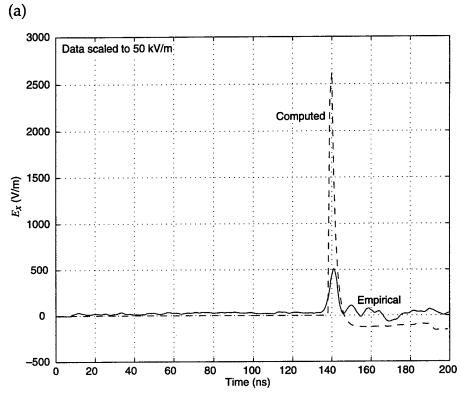


Figure B-4. Field comparison at (-332,-176,4.6) meters: (a) E_x , (b) E_y .



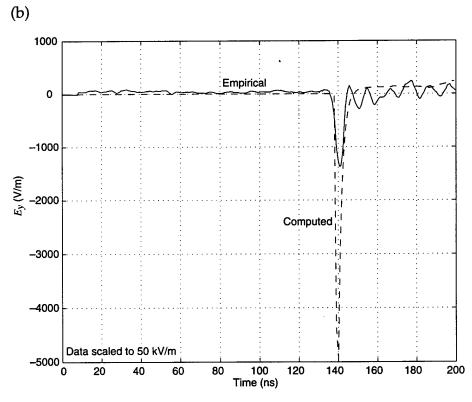
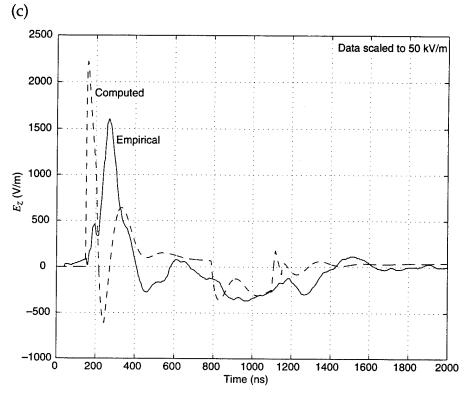


Figure B-4 (cont'd). Field comparison at (-332,-176,4.6) meters: (c) E_z , (d) H_x .



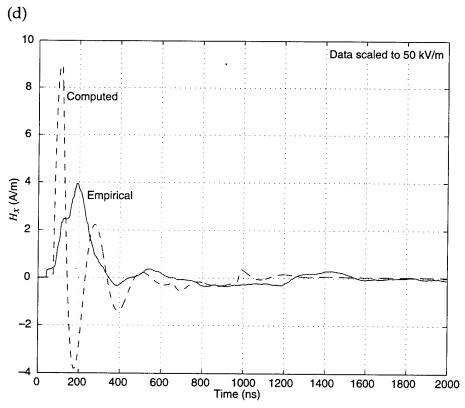
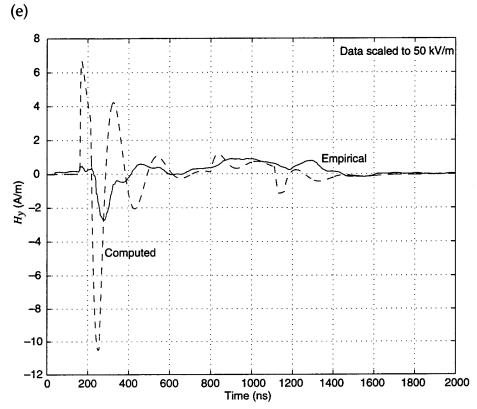


Figure B-4 (cont'd). Field comparison at (-332,-176,4.6) meters: (e) $H_{y'}$ (f) H_z .



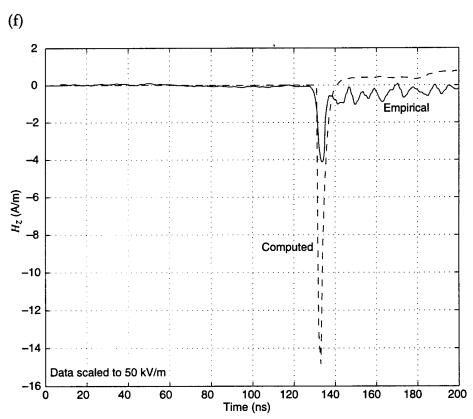


Figure B-5. Field comparison at (0,1546,4.6) meters: (a) E_x , (b) H_y .

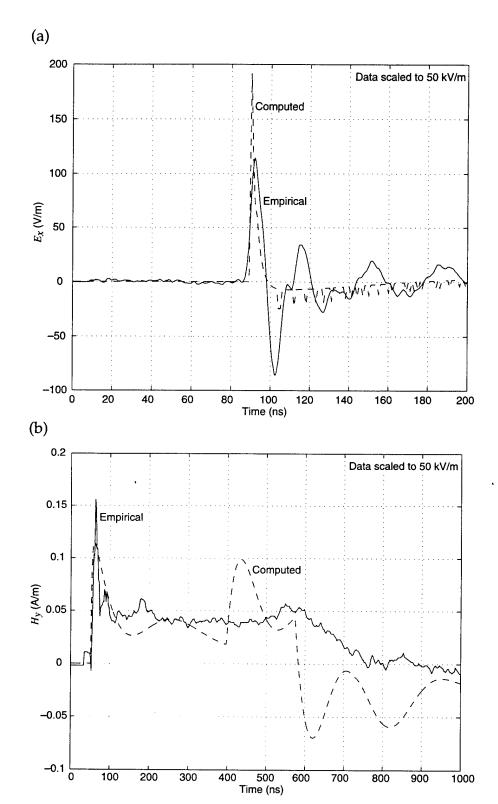


Figure B-5 (cont'd). Field comparison at (0,1546,4.6) meters: $(c) H_z$.

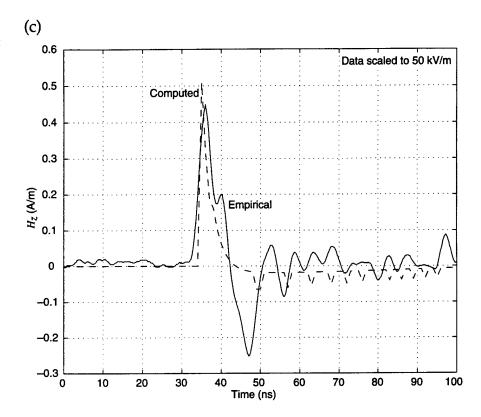
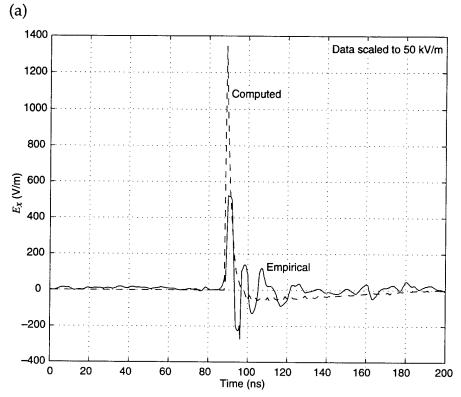


Figure B-6. Field comparison at (410,502,4.6) meters: (a) E_x , (b) E_y .



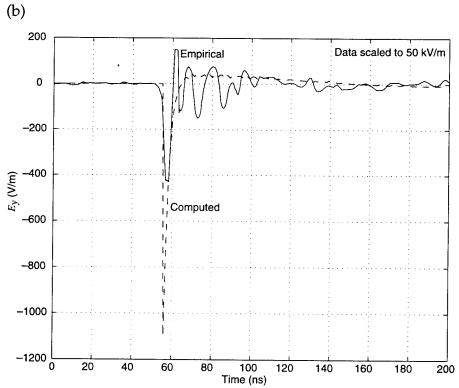
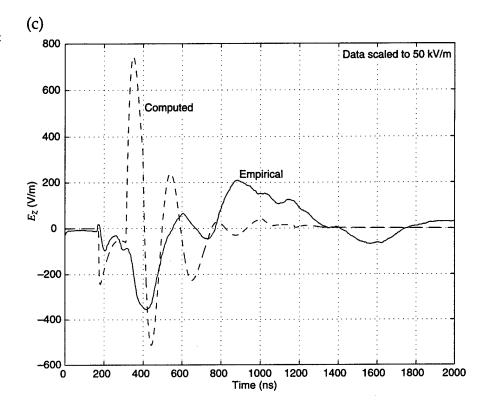


Figure B-6 (cont'd). Field comparison at (410,502,4.6) meters: (c) E_z , (d) H_x .



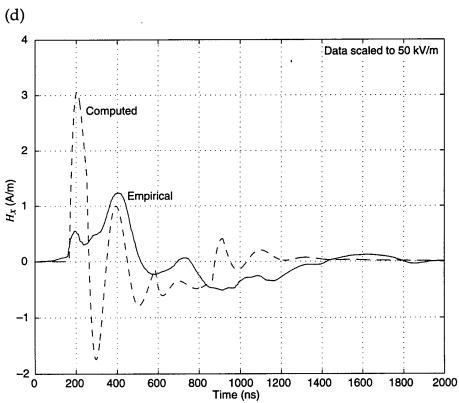
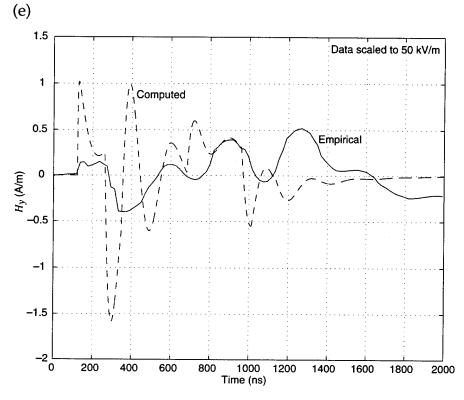


Figure B-6 (cont'd). Field comparison at (410,502,4.6) meters: (e) $H_{y'}$ (f) H_z .



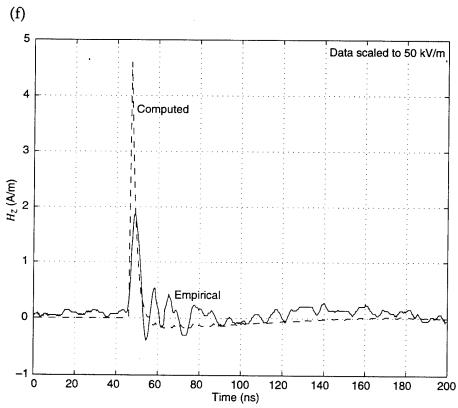


Figure B-7. Field comparison at (-444,1341,4.6) meters: (a) E_x , (b) E_y .

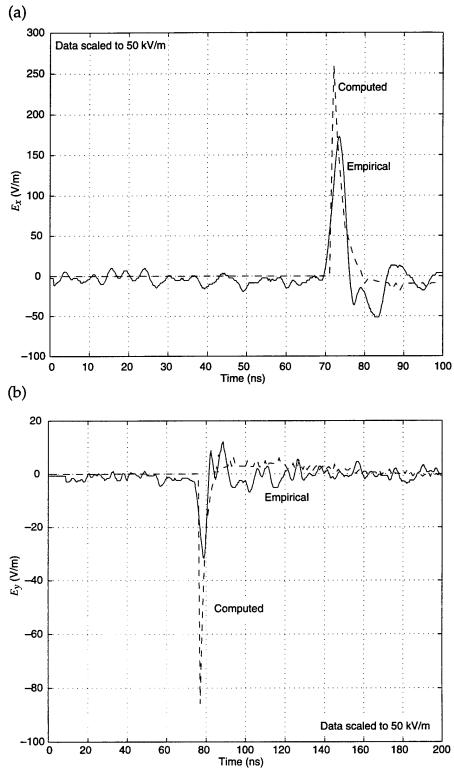
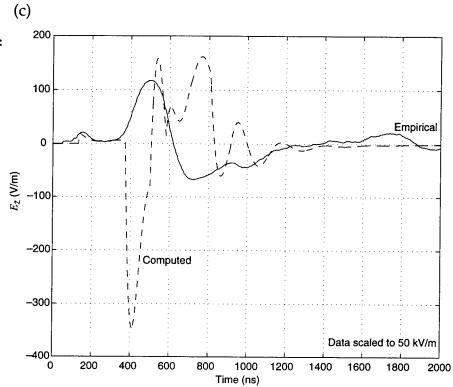


Figure B-7 (cont'd). Field comparison at (-444,1341,4.6) meters: (c) E_z , (d) H_x .



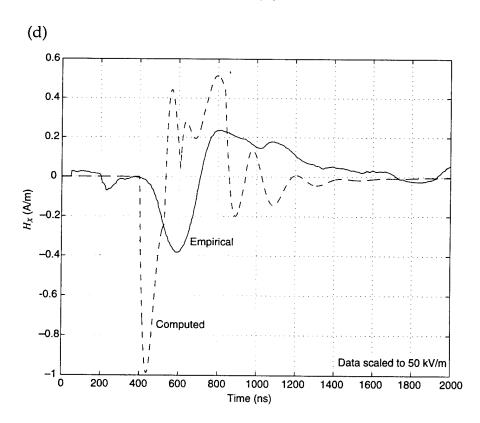
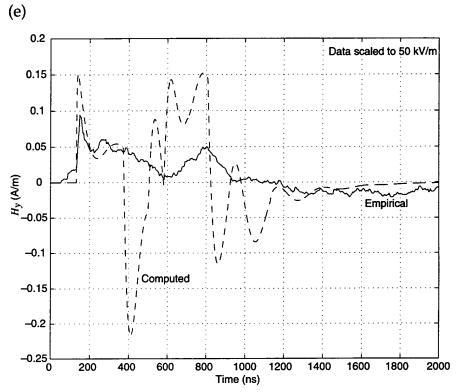


Figure B-7 (cont'd). Field comparison at (-444,1341,4.6) meters: (e) $H_{y'}$ (f) H_z .



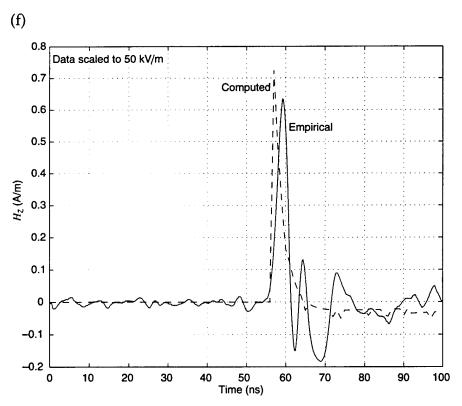
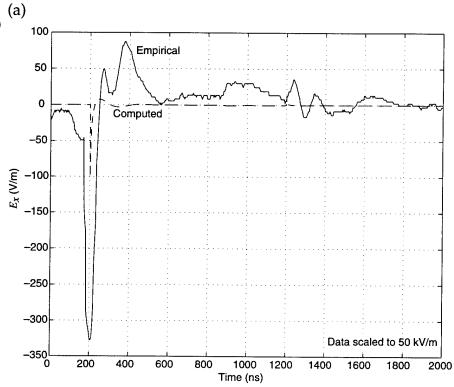


Figure B-8. Field comparison at (473,0,1) meters: (a) E_x , (b) E_z .



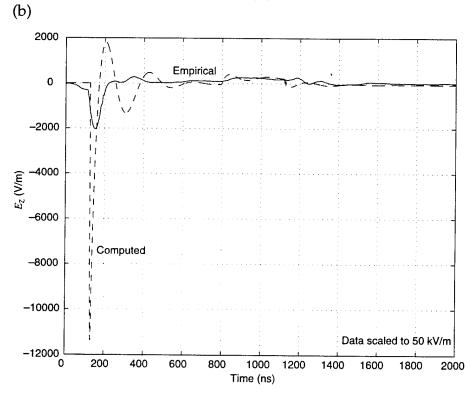


Figure B-8 (cont'd). Field comparison at (473,0,1) meters: (c) H_y .

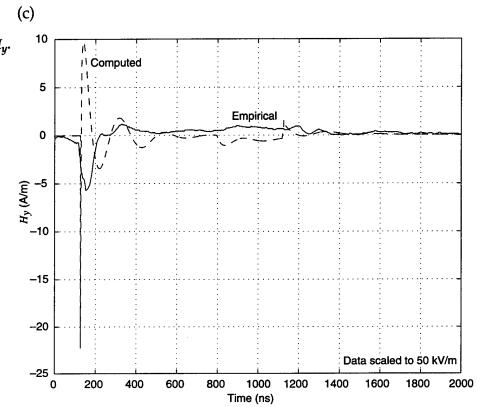
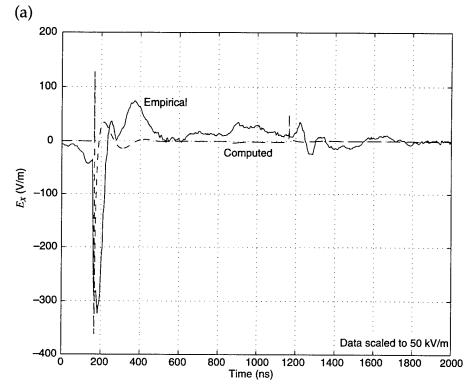


Figure B-9. Field comparison at (473,0,4.6) meters: (a) E_x , (b) E_z .



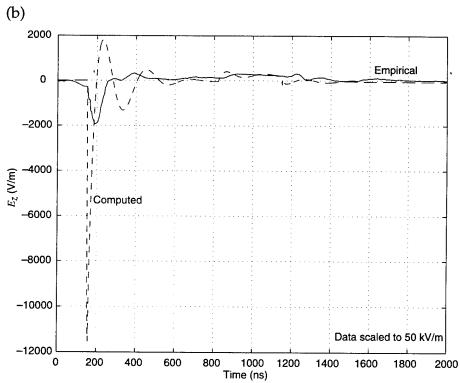
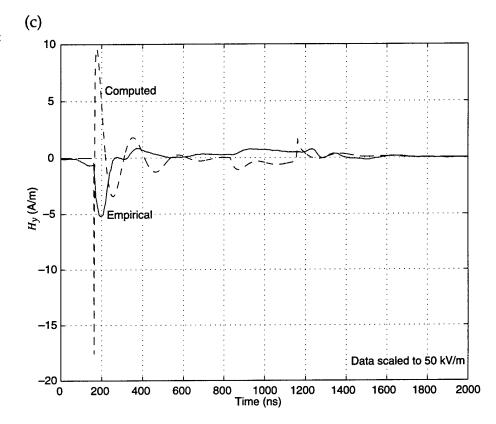


Figure B-9 (cont'd). Field comparison at (473,0,4.6) meters: (c) H_y .



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